

(19)



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(11)

EP 1 103 876 A1

(12)

EUROPEAN PATENT APPLICATION

(43) Date of publication:
30.05.2001 Bulletin 2001/22

(51) Int Cl. 7: G05B 23/02

(21) Application number: 00308311.0

(22) Date of filing: 22.09.2000

(84) Designated Contracting States:
AT BE CH CY DE DK ES FI FR GB GR IE IT LI LU
MC NL PT SE
Designated Extension States:
AL LT LV MK RO SI

(30) Priority: 26.11.1999 US 449771

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(54) Method and system for compensation of measurement error

(57) Methods and systems for reducing or removing the effect of measurement error so that an actual engine operates closer to optimum are described. In an exemplary embodiment, the method includes the steps of identifying an engine operation (e.g., acceleration) that is less than optimal (102), estimating the measurement

error associated with the less than optimal operation (104), adjusting the engine control logic based on the measurement error estimate (106), and reassessing the engine operation (108). The method is implemented, in one form, in an engine controller by programming a control processor to execute the above described method.

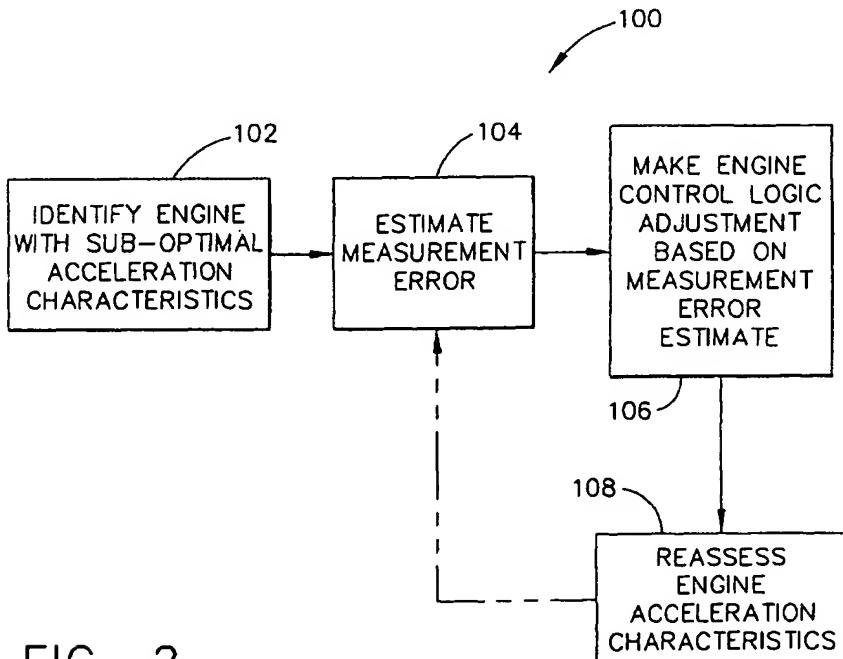


FIG. 2

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Description

[0001] The present invention relates to gas turbine engines, and more specifically, to compensating for measurement error using control logic.

[0002] Gas turbine engines include sensor and actuator position feedback used for control and diagnostics. The term sensor, as used herein, refers to engine mounted measurement devices. Values from the sensors are validated and then used implicitly by the control and diagnostic logic. The true values for the pressures, temperatures, actuator positions, speeds, and other engine variables, however, differ from the measured values. The difference between the true value and the measured value is referred to as measurement error.

[0003] Measurement errors are generally attributed to transducer inaccuracy, physical differences such as valve characteristics, and gas path profile effects. Measurement error due to transducer inaccuracy primarily results from manufacturing tolerances and signal conditioning errors. Measurement error due to valve characteristics results from geometry variations due to manufacturing tolerances and inaccuracies in valve position to flow relationships.

[0004] Measurement errors due to gas path profile effects are a combination of physical tolerances in the sensor and the sensor installation into the engine. The engine installation introduces dimensional variability, such as immersion depth and alignment, which can lead to differing sensing locations within the gas path temperature and pressure profile. Profile measurement error is usually much larger at aft stages of an engine since combustion and gas mixing are significant factors in such measurement error.

[0005] An absolute level of measurement error depends on an accuracy specification used in device design, the type of device, and the device utilization. For example, a pressure transducer critical for engine control typically must be much more accurate than a condition monitoring sensor for optional equipment that is not flight critical. Once installed, measurement error often is assumed to be consistent with respect to sign (i.e., positive or negative error). However, error magnitude will vary with operating conditions, e.g., a larger measurement error at high power versus idle.

[0006] Since an engine controller uses indicated values from the sensors, the effects of measurement error are accommodated in an overall engine control law design process. Typically, control laws are designed to include margins for a worst-case measurement error. While this approach provides a safe margin of operation for a worst case engine, this approach also results in larger than necessary margins for all other engines.

[0007] The present methods and systems, in one aspect, reduce conservatism inherent in selecting a worst-case measurement error by reducing, or removing, the effect of measurement error so that an actual engine operates closer to optimum. In an exemplary embodiment,

the method includes the steps of identifying an engine operation (e.g., acceleration) that is less than optimal, estimating the measurement error associated with the less than optimal operation, adjusting the engine control logic based on the measurement error estimate, and re-assessing the engine operation. The method is implemented, in one form, in an engine controller by programming a control processor to execute the above described method.

5 [0008] The above described method provides the advantage that measurement error adjustments are made depending upon the characteristics of each engine rather than making a same adjustment for all engines of a particular type or model. In addition, an engine with below nominal operating characteristics can be returned to nominal operation without replacing sensors or other related hardware, which results in increased time on-wing, lower acceleration time, and reduced operating temperatures.

10 [0009] An embodiment of the invention will now be described, by way of example, with reference to the accompanying drawings, in which:

25 Figure 1 is a block diagram of an engine control system; and

Figure 2 is a process flow diagram for performing measurement error estimation adjustment in the engine control system shown in Figure 1.

30 [0010] Figure 1 is a block-diagram of an engine control system 10. As shown in Figure 1, multiple sensors (Sensor 1 - Sensor N) 12 are coupled to an engine controller 14. Controller 14, in one embodiment, is a processor programmed to execute control logic for generating control signals 16. More specifically, controller 14 generates engine control signals based on the measured values supplied by sensors 12. Sensors 12 include, for example, a fan speed sensor (N1), a fan inlet temperature (T12) sensor, a compressor inlet temperature (T25) sensor, a high pressure compressor discharge pressure (P3) transducer, fuel metering valve (FMV) position sensor, as well as many other sensors. Such sensors are well known in the art.

35 [0011] As explained above, engine controller 14 uses indicated values from sensors 12 to generate control signals. The control laws used by controller 14 in generating such signals typically include margins for a worst-case measurement error. While this approach provides a safe margin of operation for a worst case engine, this approach also results in larger than necessary margins for all other engines.

40 [0012] Generally, the present methods and systems are directed towards reducing the conservatism inherent in selecting a worst-case measurement error by reducing or removing the effect of measurement error so that the actual engine operates closer to optimum. The method includes the steps of identifying an engine operation that is less than optimal, estimating the meas-

urement error associated with the less than optimal operation, adjusting the engine control logic based on the measurement error estimate, and reassessing the engine operation. The method is implemented, in one form, in controller 14. Specifically, in the one embodiment, the processor in controller 14 is programmed to execute the above described processing.

[0013] More specifically, since the measurement error manifests itself in engine operational effects, these operational effects are utilized in the measurement error estimation. For example, time to accelerate an engine is a function of the transient regulator. The measurement error effects of the sensors utilized in the transient regulator logic impacts the actual acceleration time. If an actual acceleration response is measured and compared to a nominal (no error) response, then an estimate of the overall measurement error on the transient regulator can be made, i.e., measured value minus nominal value. After generating a measurement error estimate, then the appropriate sensor can be made "error free" by making a compensation adjustment in the engine control logic.

[0014] Figure 2 illustrates a process 100 for compensating for measurement error in the engine control logic if less than optimal acceleration operation is detected. Referring specifically to Figure 2, controller 14 is programmed to determine whether the engine is exhibiting slow acceleration characteristics by comparing the measured acceleration characteristics from sensors 12 to nominal values, e.g., pre-stored in a memory of controller 14: If the difference between the measured and nominal values exceeds a predetermined threshold value, then the engine characteristics are considered sub-optimal 102 and controller then proceeds with estimating the overall transient regulator measurement error 104.

[0015] To estimate the overall measurement error, a large number of simulated engines with random or specified measurement error are defined. Simulated transients are run over a specified operating regime (altitude, ambient temperature, bleed settings, etc.) for different engine quality and deterioration levels. A multi-variate regression fit is used to estimate the measurement error as a function of sensor data and operational data, such as regulator usage (i.e., which regulator is in control and for how long). The regression fit is obtained by one of the techniques well known to those skilled in the art, such as linear regression, response surface fits using polynomials or neural networks. In one specific embodiment, instead of estimating each measurement error individually, an equivalent fuel metering valve (FMV) bias or measurement error is estimated using the regression fit.

[0016] After generating a measurement error estimate, the appropriate sensor value used in controller 14 is made "error free" by making a compensation adjustment in the engine control logic 106. Specifically, the measured value is adjusted by a percentage (0 - 100%)

of the estimated measurement error. The percentage selected can be equal to 100%, or to a lower value, such as 75%, to allow for inaccuracies in estimating the measurement error, and can be pre-stored as an adjustable constant in the controller memory. The adjusted sensor value is then used by controller 14 in further processing and in generating control signals.

[0017] Once the adjustment is made, the engine acceleration characteristics are reassessed 108. If the engine still exhibits sub-optimal acceleration characteristics additional measurement error adjustments are made by returning to step 102 and then proceeding with processing.

[0018] The above described methods and systems provide the advantage that measurement error adjustments are made depending upon the characteristics of each engine with its specific sensor set and physical device (valves, etc.) characteristics rather than making a same adjustment for all engines of a particular type or model. In addition, an engine with below nominal operating characteristics can be returned to nominal operation without replacing sensors or other related hardware, which results in increased time on-wing, lower acceleration time, and reduced operating temperatures.

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Claims

1. A method for controlling operation of an engine, said method comprising the steps of:
 - identifying an engine operation that is less than optimal;
 - estimating a measurement error (104) associated with the less than optimal engine operation; and
 - adjusting engine control logic (106) based on the measurement error estimate.

2. A method for controlling acceleration of an engine, said method comprising the steps of:
 - determining whether the engine acceleration is slow (102);
 - if the engine acceleration is slow, then:
 - estimating an overall transient regulator measurement error (104); and
 - adjusting engine control logic (106) based on the overall transient regulator measurement error estimate.

3. A method in accordance with Claim 1 or claim 2 further comprising the step of reassessing the engine operation or the engine acceleration characteristics (108) after adjusting the engine control logic (106).

4. A method in accordance with Claim 1 or claim 2 wherein identifying if engine operation or engine acceleration is less than optimal comprises the step

of comparing a measured sensor values to nominal values.

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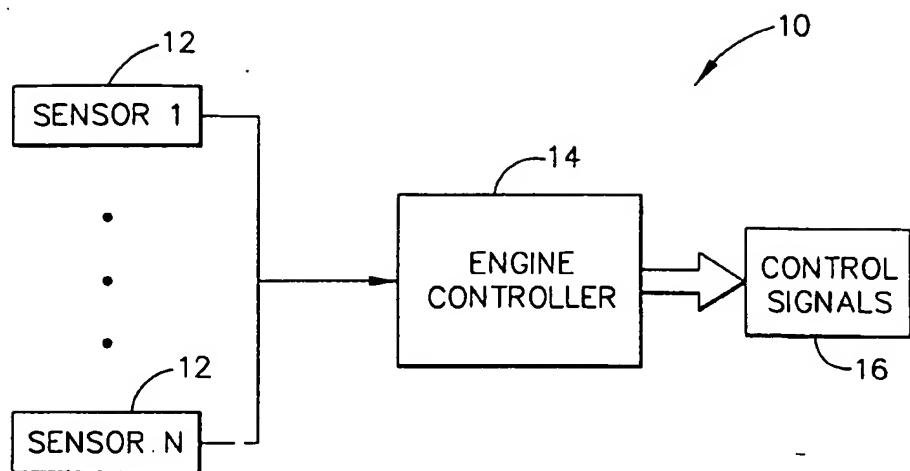


FIG. 1

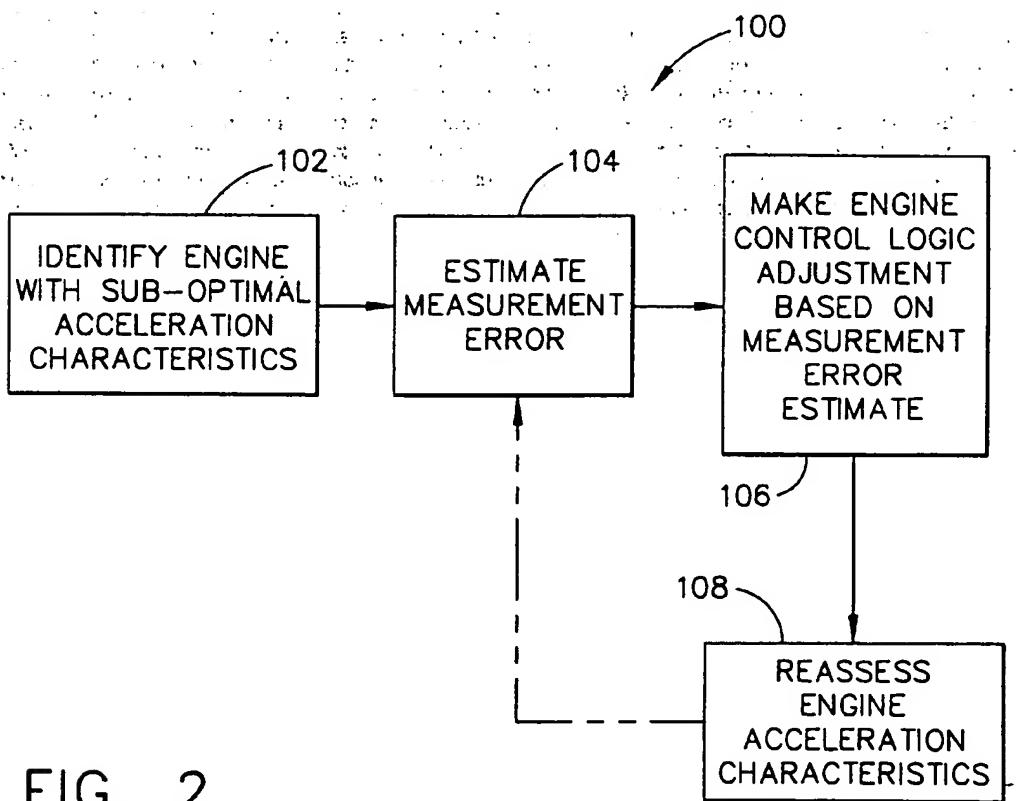


FIG. 2



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EUROPEAN SEARCH REPORT

Application Number
EP 00 30 8311

DOCUMENTS CONSIDERED TO BE RELEVANT			CLASSIFICATION OF THE APPLICATION (Int.Cl.)
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim	
X	US 5 625 143 A (KADOTA YOICHI) 29 April 1997 (1997-04-29) * claim 1 *	1,7	G05B23/02
X	US 5 394 331 A (DUDEK KENNETH P ET AL) 28 February 1995 (1995-02-28) * column 2, line 39 - column 4, line 44 *	1,7	
A	EP 0 744 608 A (MITSUBISHI MOTORS CORP) 27 November 1996 (1996-11-27) * column 7, line 22 - column 9, line 1 *	1,7	
A	US 5 834 624 A (NAKAGAWA NORIHISA) 10 November 1998 (1998-11-10) * column 9, line 14 - line 28; figure 4 *		
A	US 4 423 594 A (ELLIS STANLEY H) 3 January 1984 (1984-01-03) * column 9, line 5 - column 13, line 10 *		
A	US 5 377 536 A (ANGERMAIER ANTON ET AL) 3 January 1995 (1995-01-03) * column 6, line 23 - line 60 *		TECHNICAL FIELDS SEARCHED (Int.Cl.) G05B
The present search report has been drawn up for all claims			
Place of search	Date of completion of the search	Examiner	
THE HAGUE	16 February 2001	Kelperis, K	
CATEGORY OF CITED DOCUMENTS		T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons & : member of the same patent family, corresponding document	
<small>EPO FORM 1503.03.02 (P04/01)</small>			

**ANNEX TO THE EUROPEAN SEARCH REPORT
ON EUROPEAN PATENT APPLICATION NO.**

EP 00 30 8311

This annex lists the patent family members relating to the patent documents cited in the above-mentioned European search report.
The members are as contained in the European Patent Office EDP file on
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16-02-2001

Patent document cited in search report		Publication date		Patent family member(s)	Publication date
US 5625143	A	29-04-1997	JP	8028340 A	30-01-1996
US 5394331	A	28-02-1995	US	5293553 A	08-03-1994
			US	5270935 A	14-12-1993
			US	5273019 A	28-12-1993
EP 0744608	A	27-11-1996	JP	2795062 B	10-09-1998
			JP	5340294 A	21-12-1993
			JP	2836421 B	14-12-1998
			JP	6229310 A	16-08-1994
			EP	0744609 A	27-11-1996
			EP	0744610 A	27-11-1996
			AU	660823 B	06-07-1995
			AU	4354993 A	04-01-1994
			DE	69313814 D	16-10-1997
			DE	69313814 T	09-04-1998
			EP	0609451 A	10-08-1994
			WO	9325811 A	23-12-1993
			KR	140685 B	01-07-1998
			US	5506778 A	09-04-1996
US 5834624	A	10-11-1998	JP	9329575 A	22-12-1997
US 4423594	A	03-01-1984		NONE	
US 5377536	A	03-01-1995	EP	0583496 A	23-02-1994
			DE	59204440 D	04-01-1996
			JP	6185400 A	05-07-1994